

PowerPoint 版

Torsion Type Hydraulic Gates

# **Mitigation of Warping & Optimum Design**

*T e r a M a t s u*

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## 1 . Introduction

A torsion type closed thin shell section is overwhelmingly superior in structural functions to a bending type structure because the torsion type resists against external load with second power of the closed area whereas the bending type resists with the sectional moment of inertia of thin shell, and this superiority gets more remarkable as the structural span becomes longer.

We have two theories in structural torsion, the simple torsion and the bending-torsion. In case of the simple torsion, shearing stress yields on a cross section due to simple torsion moment. In case of the bending-torsion, bending-torsion moment is applied to the section in addition to simple torsion moment. Shearing stress due to simple torsion moment distributes uniformly over the section whereas shearing stress due to bending-torsion moment meanders so that peak value of their total sometimes become much more than 200% of simple torsion alone. Furthermore, vertical stress that is causally related to sectional warping and in equilibrium with shearing stress of bending-torsion arises on the section.

Since stress is in proportion to the product of a form coefficient, deformation and a spring constant that shearing and vertical stresses of bending-torsion can decrease or be eliminated with form coefficient reduction (patent pending). On the other hand, the form coefficient reduction will result in section modulus decrease which may cancel its effect because intensity of deformation is equal to internal force divided by the product of a section modulus and a spring constant.

The most important section modulus controlling deformation of a torsion type structure is  $J_t$ . The product of  $J_t$  and a spring constant is a sectional rigidity which resists simple torsion moment. Although  $J_t$  decreased because of the form coefficient reduction, gate weight cut is possible by making up the lost  $J_t$  with help of cross section form change.

Superiority of the torsion type structure to the bending type may be made more certain by applying the form coefficient operation to **optimum design** whose object function is cost.

2 . What is bending-torsion?

2 . 1 Who started to use the name of "Bending-torsion"?

Use of the name "Bending-torsion" was started by H. Wagner. He established general theory for bending-torsion of H shape shown on Fig.-2-1. In his study,  $C_{bt}$  (section modulus for bending-torsion) was called, Bieguugsverdrehtungswiderstand (a sectional rigidity of bending-torsion).

In Fig.-2-1, End A is a support side of the H shape and End B is supposed to be twisted around X axis by  $T$  due to external torsion moment and section 1234 of End B warps and gets to 1'2'3'4'. The warping will change along X axis if any restriction or change in torsion moment exists between A and B. Existence of change in warping means each flange of H shape bends in its own plane so that a pair of shear forces exists on the flange sections as shown on Fig.-2-2 and it forms a couple. This couple together with simple torsion moment is in balance with external torsion moment.

It is reported that "Bending-torsion" was named after above explained structural phenomena. The couple is called bending-torsion moment. There are many cases which give different images from such as the original explanation since study in this field spread but still this name is surviving.

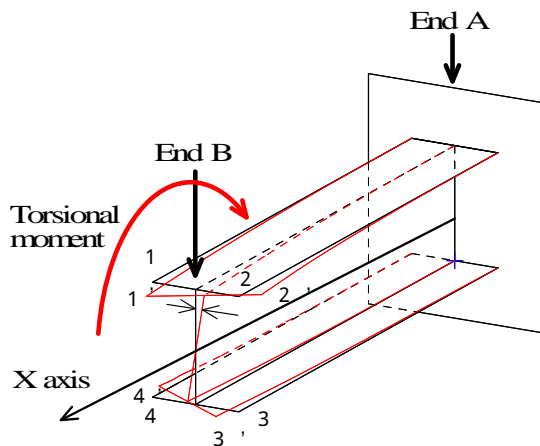


Fig.-2-1 Origin of "Bending-torsion"

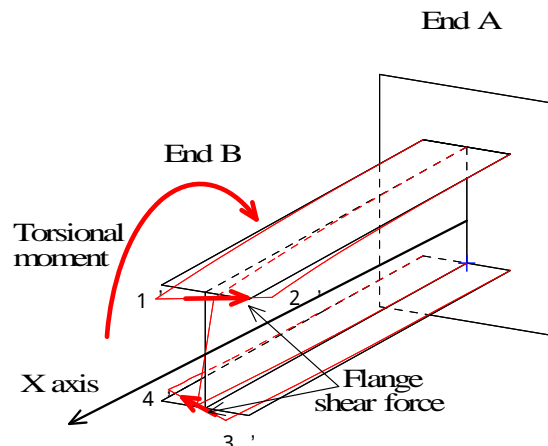


Fig.-2-2 Flange shearing stress

2 . 2 Simple torsion and bending-torsion

Explanation of difference between the simple torsion and the bending-torsion is

presented on a closed section of thin shell as an example that has a different image of "Bending-torsion".

( 1 ) The simple torsion

Fig.-2-3 shows an example of deformation due to the simple torsion of a closed thin shell section. End A is a support side and End B is a loading side of the closed section, which is twisted around X axis due to torsion moment applied on End B and section 1234 of End B is transformed to 1'2'3'4' without sectional warping. The simple torsion is a twisting without sectional warping. It is also called uniform twisting. It includes not only twisting without warping (for instance, round section) but also twisting having warping but no bending-torsion and twisting whose bending-torsion is negligible. Most of twisting is deemed to belong to this category and all of them are handled as if they have no warping. Internal forces on a cross section of simple torsion are simple torsion moment, bending moment and shearing force in general.

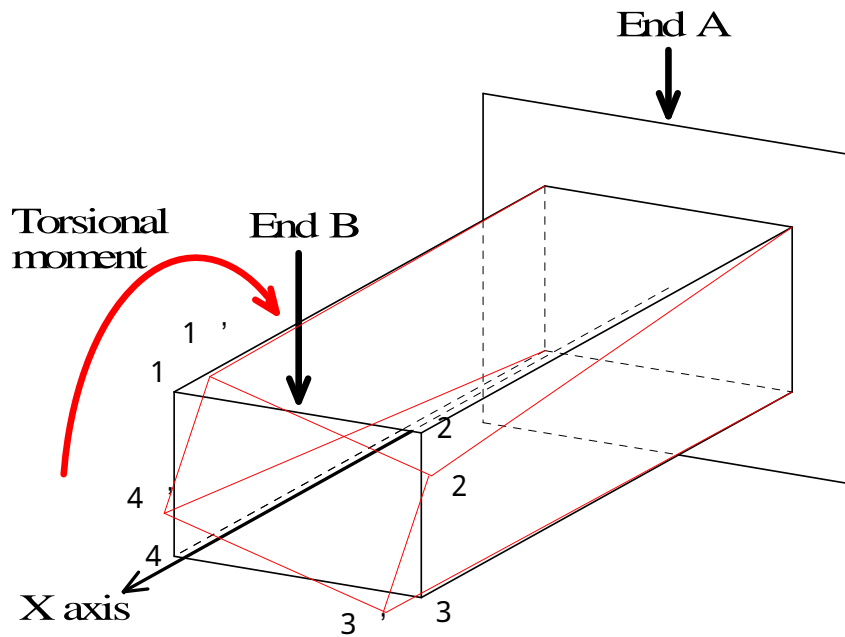


Fig.-2-3 Simple torsion of a closed section of thin shell

( 2 ) The bending-torsion

Fig.-2-4 shows deformation of a rectangular closed section of thin shell in bending-torsion. End A is a support side and End B is supposed to be twisted around X

axis due to external torsion moment and section 1234 of End B warps and gets to 1'2'3'4'. The warping will change along X axis due to restrictions or change in torsion moment existing between A and B. As is seen on Fig.-2-2, shearing forces yield on the bottom and the top plate and on the both side plates and they form two pairs of coupling. These couples together with simple torsion moment are in balance with external torsion moment. In addition to the three internal forces of simple torsion, two other items, bending-torsion moment and vertical stress of sectional warping, exist on the cross section in case of the bending-torsion.

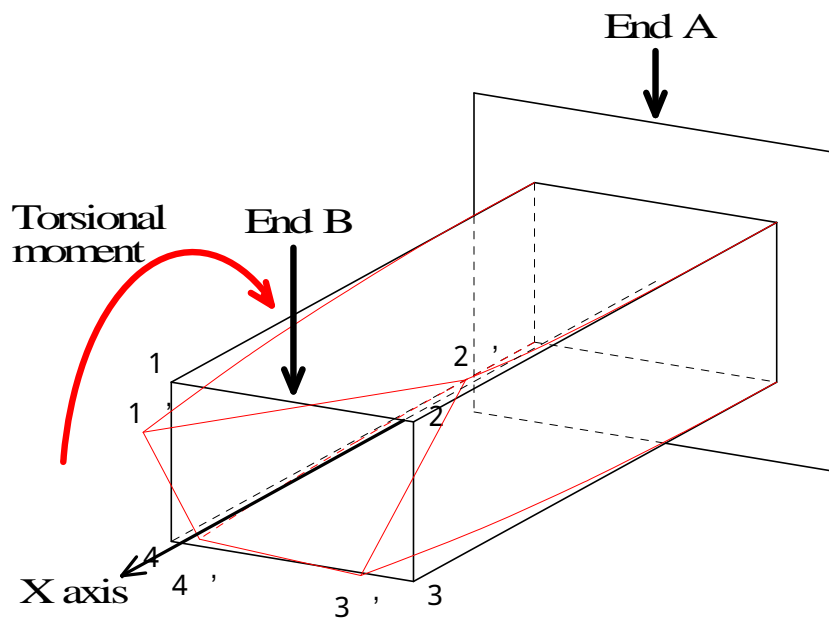


Fig.-2-4 Bending-torsion of a closed section of thin shell

### 3 . Effect of the bending-torsion

Although there is no big change in internal forces, the shearing stress which distributes uniformly over a section in case of simple torsion will meanders due to existence of bending-torsion moment and its peak value increase to a remarkable extent sometimes that is shown in following examples. The elastic equation method shown on bibliography (6) was applied to the structural analysis of the examples.

#### 3 . 1 Effect on internal force

Fig.-3-1 is Fundamental case of a rectangular section, on which it will be shown that no big change due to the bending-torsion occurs in internal forces of simple torsion. 0 of the front view in the figure is a support end and 8 is a free end.

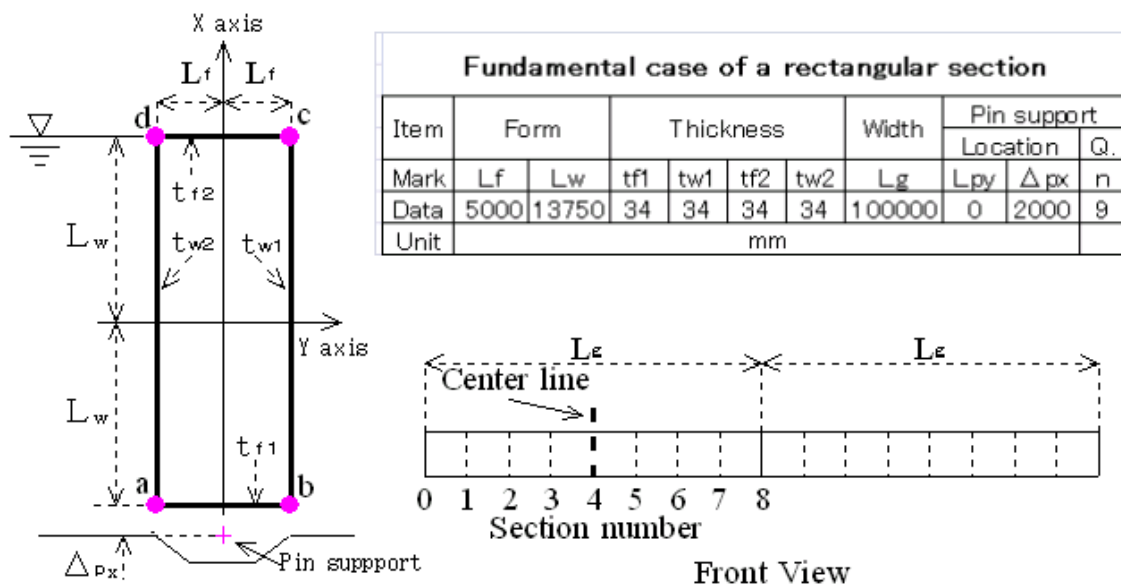


Fig.-3-1 Fundamental case of a rectangular section

Fig.-3-2 and Fig.-3-3 are internal force analyzed by the simple torsion theory and Fig.-3-4 thru Fig.-3-6 are internal force and deformation analyzed by the bending-torsion theory. Deformation of the simple torsion theory is shown in Fig.-3-4 for the bending-torsion theory. My which is bending moment around Y axis and  $Q \times$  which is shearing force in X direction are zero because the pin support in Fig.-3-1 locates on X axis on which shearing center is also.

Change in cross sectional internal forces due to additional bending-torsion is as

follows.

- (1) Internal torsion moment has small difference, but no change at support terminal.
- (2) Sum of simple and bending torsion moments is equal to internal torsion moment.
- (3) Small decreases in bending moment and shearing force on the section.
- (4) Twisting angle has small decrease, but no change at free terminal.

( 1 ) Analysis of the simple torsion theory

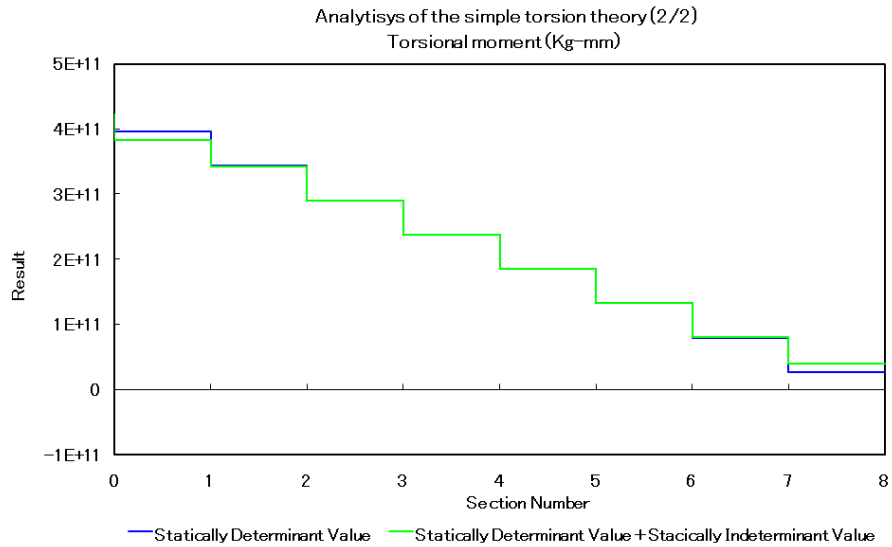


Fig.-3-2 Analysis of the simple torsion theory • • torsion moment

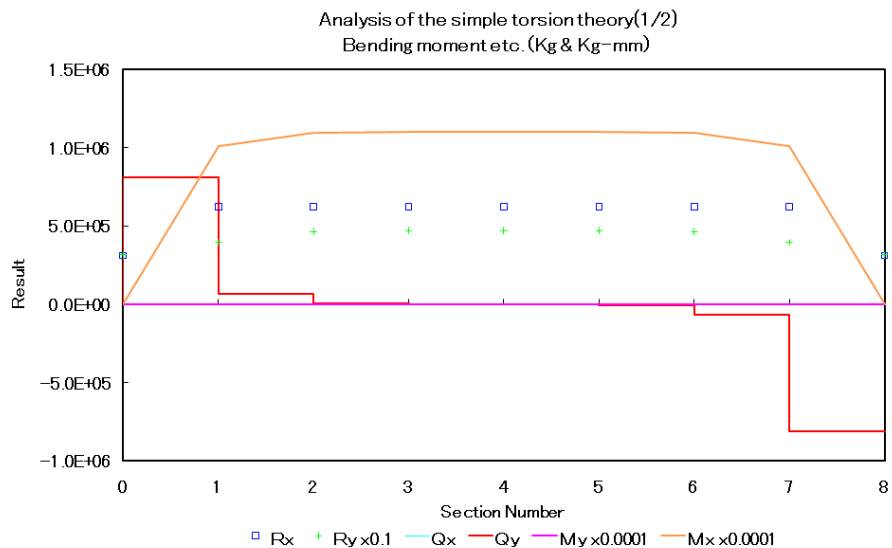


Fig.-3-3 Analysis of the simple torsion theory • • Bending moment etc.

( 2 ) Analysis of the bending-torsion theory

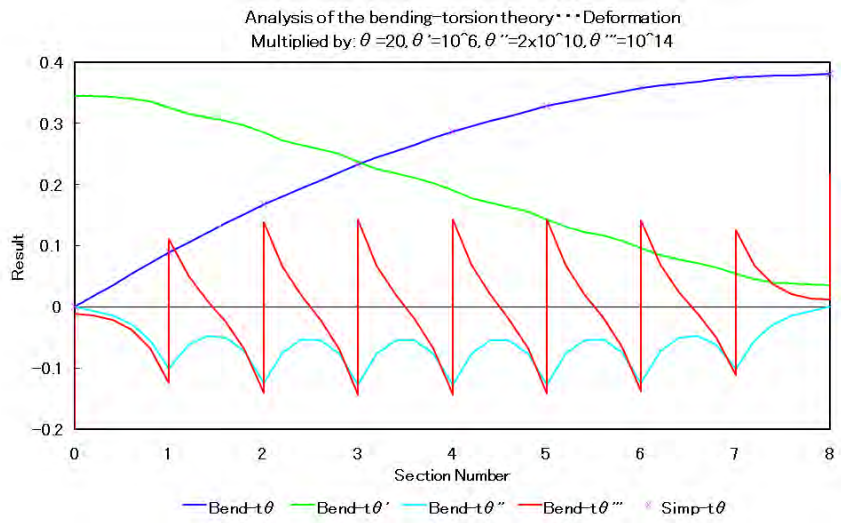


Fig.-3-4 Analysis of the bending-torsion theory • • Deformation

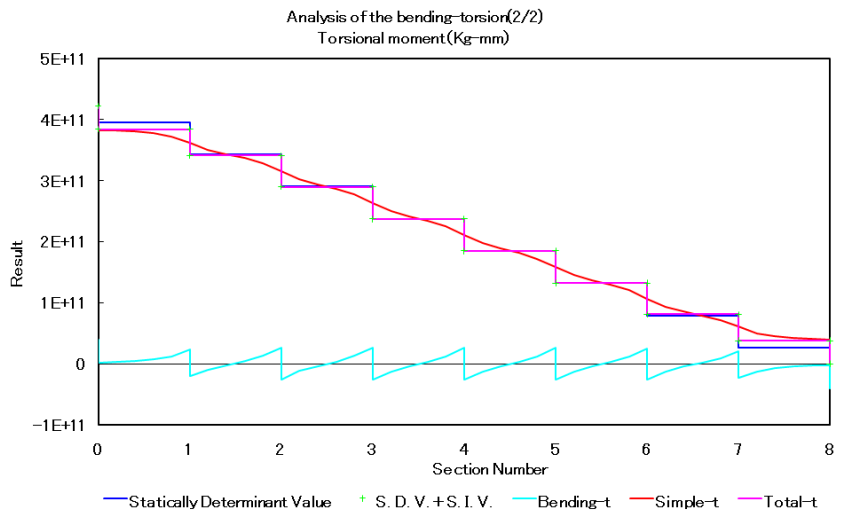


Fig.-3-5 Analysis of the bending-torsion theory • • torsion moment

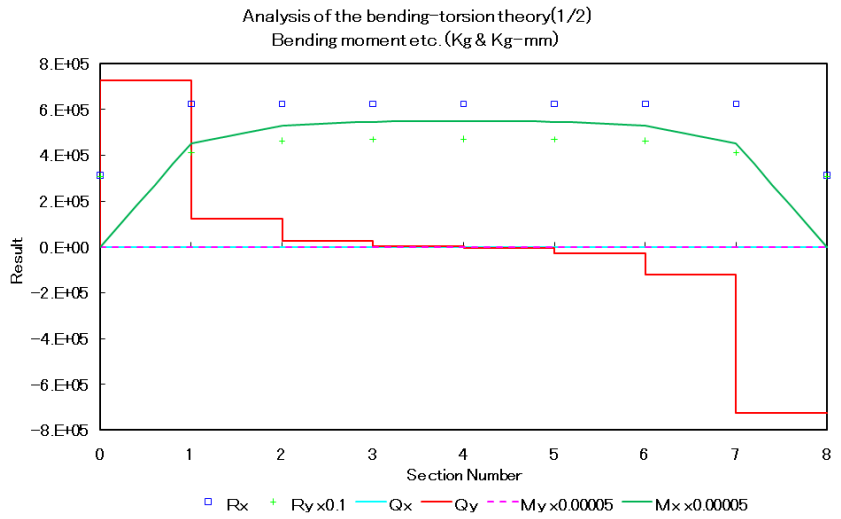


Fig.-3-6 Analysis of the bending-torsion theory • • Bending moment etc.

### 3 . 2 Effect on stress

It is observed on both the fish belly section and the rectangular section that the shearing stress which distributes uniformly over a section of the simple torsion will meanders due to additional bending-torsion moment and its peak value increases to a remarkable extent. The section whose stress is shown is identified by, for example, 1out or 1in where 1 indicates section number and out or in indicates outside or inside of the section with respect to a center line of gate. Stresses of both side of a section are presented since sign of bending-torsion moment turns over at the section. The section number and the center line are shown on a corresponding figure.

#### 3 . 2 . 1 Stress on a fish belly section

A fish belly section is often utilized for reservoir or river flaps. Fig.-3-7 shows its example. Section Number 0 on the figure is a support end and that of 8 is a free end.

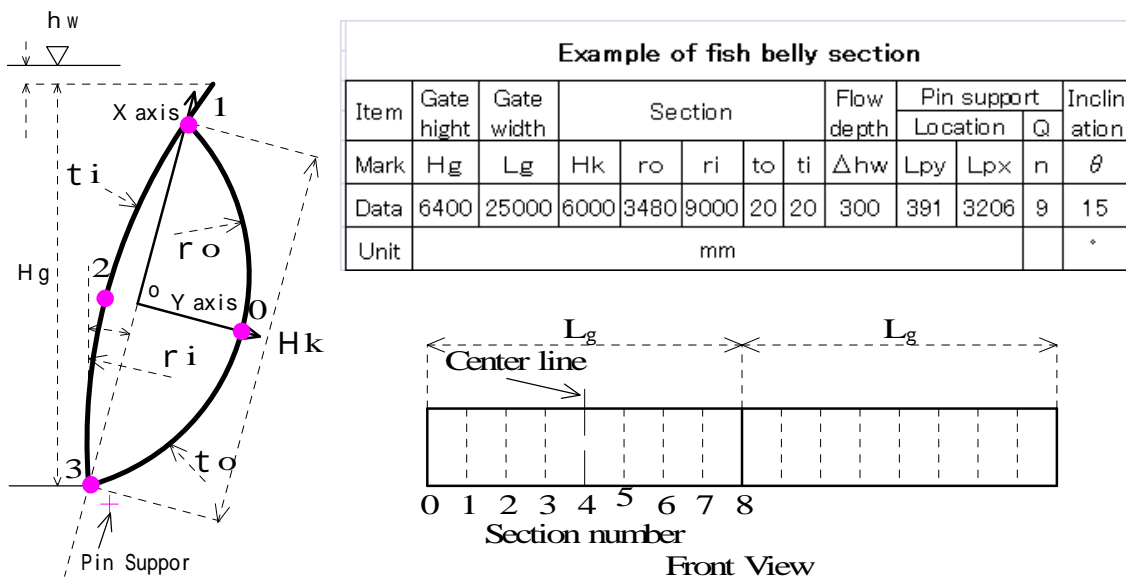


Fig.-3-7 Example of a fish belly section

Fig.-3-8 shows sectional stress of the simple torsion theory and Fig.-3-9 shows that of the bending-torsion theory. Lateral axis on the graphs is girth length along the section shown on Fig.-3-7 and the numbers 0, 1, 2 and 3 on the axis correspond to pink colored points on the figure.

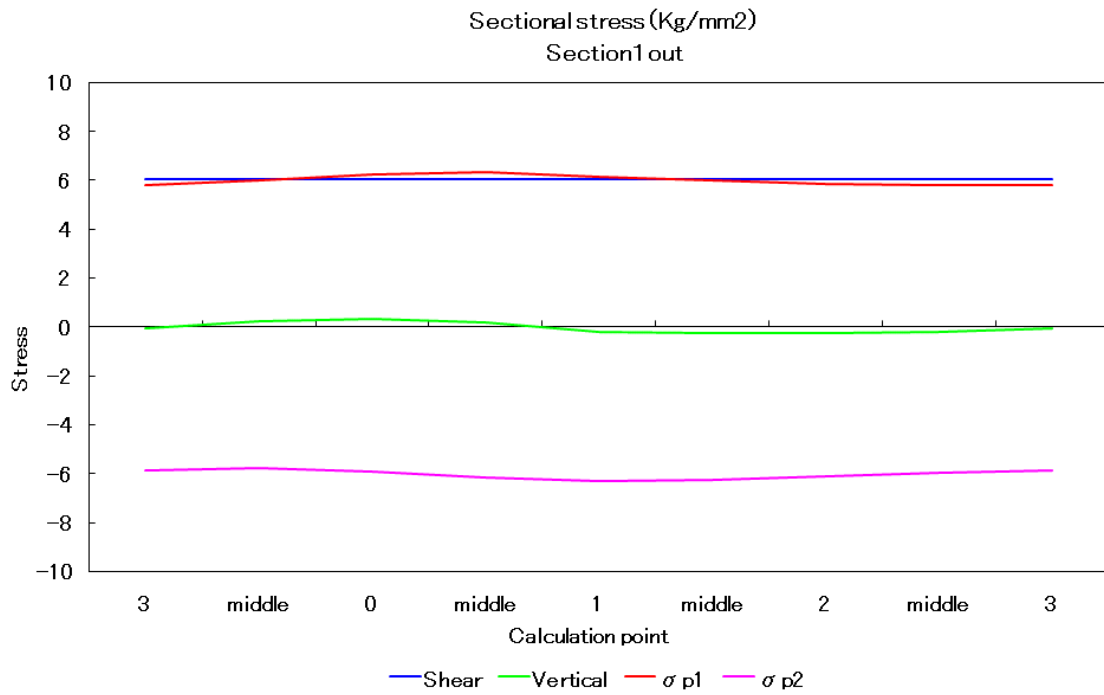


Fig.-3-8 Sectional stress of the simple torsion theory (fish belly section)

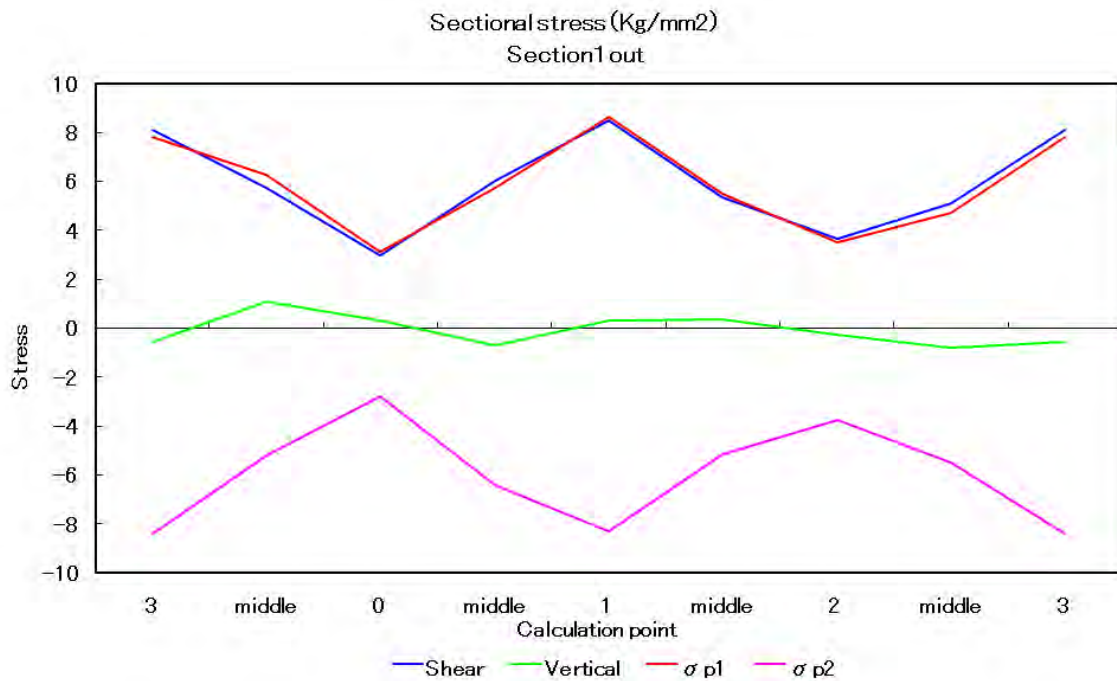


Fig.-3-9 Sectional stress of the bending-torsion theory (fish belly section)

**Explanation of Fig.-3-8 (simple torsion):** Shear is abbreviation of shearing stress which corresponds to the sum of simple torsion stress and shearing stress relating to bending. Vertical is abbreviation of vertical stress which corresponds the sum of stresses by bending moments around X and Y axes.  $p_1$  and  $p_2$  are principal stresses calculated from the shearing stress and the vertical stress. Stress status of the section is almost in pure shearing because absolute value of  $p_1$  and  $p_2$  is almost equal to the shearing stress.

**Explanation of Fig.-3-9 (bending-torsion):** Shear is abbreviation of shearing stress which corresponds to the sum of simple torsion stress, bending-torsion stress and shearing stress relating to bending. Vertical is abbreviation of vertical stress which corresponds the sum of vertical stress by sectional warping and stresses by bending around X and Y axes.  $p_1$  and  $p_2$  are principal stresses calculated from the shearing stress and the vertical stress. Stress status of the section is almost in pure shearing because absolute value of  $p_1$  and  $p_2$  is almost equal to the shearing stress. Vertical stress by sectional warping corresponds to majority of total vertical stress but is not still dominant among sectional stresses. The more calculation points, the more smooth curves we will get in cases of the shearing stress and the principal stresses. The maximum shearing stress increase rate of bending-torsion versus simple torsion is about 33 %.

### 3 . 2 . 2 Stress on a rectangular section

A rectangular section has been applied to a flap gate of ultra-large ship repair docks. This sectional form is recommendable for a torsion type flood gate of our technical proposal. Two examples are presented for comparison. One shows stress on a section in the vicinity of a support end where internal twisting moment is big and other shows stress on a section in the vicinity of a free end where internal twisting moment is comparatively small.

#### 3 . 2 . 2 . 1 Stress in the vicinity of a support end

Sectional stress in the vicinity of a support end of Fundamental case shown on Fig.-3-1 is presented. Applied twisting moment is maximum at a support end.

##### ( 1 ) Section Iout

Fig.-3-10 shows sectional stress of the simple torsion theory and Fig.-3-11 shows that of the bending-torsion theory. Lateral axis on the graphs is girth length along the section shown on Fig.-3-1 and the sign a, b, c and d in the graph correspond to pink colored points on the figure.

**Explanation of Fig.-3-10 (simple torsion):** Shear is abbreviation of shearing stress which corresponds to the sum of simple torsion stress and shearing stress relating to bending. Vertical is abbreviation of vertical stress which corresponds the sum of stresses by bending around X and Y axes.  $p_1$  and  $p_2$  are principal stresses calculated from the shearing stress and the vertical stress. Stress status of the section is almost in pure shearing because absolute value of  $p_1$  and  $p_2$  is almost equal to the shearing stress.

**Explanation of Fig.-3-11 (bending-torsion):** Shear is abbreviation of shearing stress which corresponds to the sum of simple torsion stress, bending-torsion stress and shearing stress relating to bending. Vertical is abbreviation of vertical stress which corresponds the sum of vertical stress by sectional warping and stresses by bending around X and Y axes.  $p_1$  and  $p_2$  are principal stresses calculated from the shearing stress and the vertical stress. Stress status of the section is almost in pure shearing because absolute value of  $p_1$  and  $p_2$  is almost equal to the shearing stress. The warping stress which corresponds to majority of the vertical stress may affect sectional stress to certain extent but is not dominant. The maximum shearing

stress increase rate of bending-torsion versus simple torsion is about 20 %. Increase of stress occurs between a and b (bottom plate) and c and d (top plate), and decrease occurs between b and c and d and a (side plates).

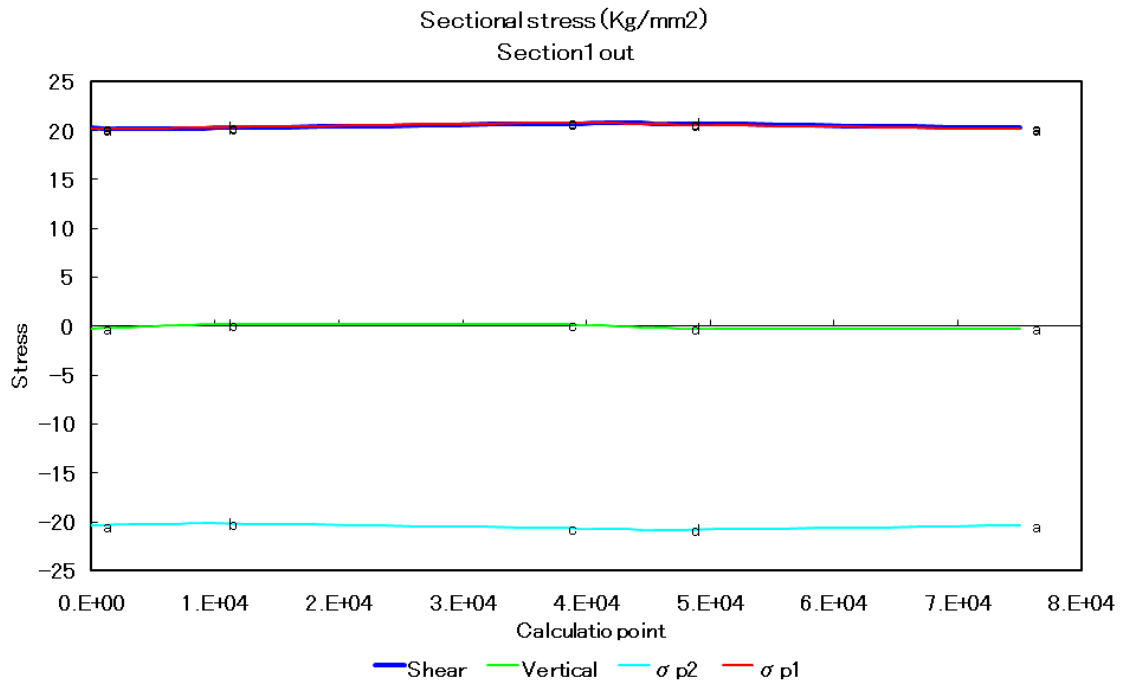


Fig.-3-10 Sectional stress of the simple torsion theory (Rectangular section)

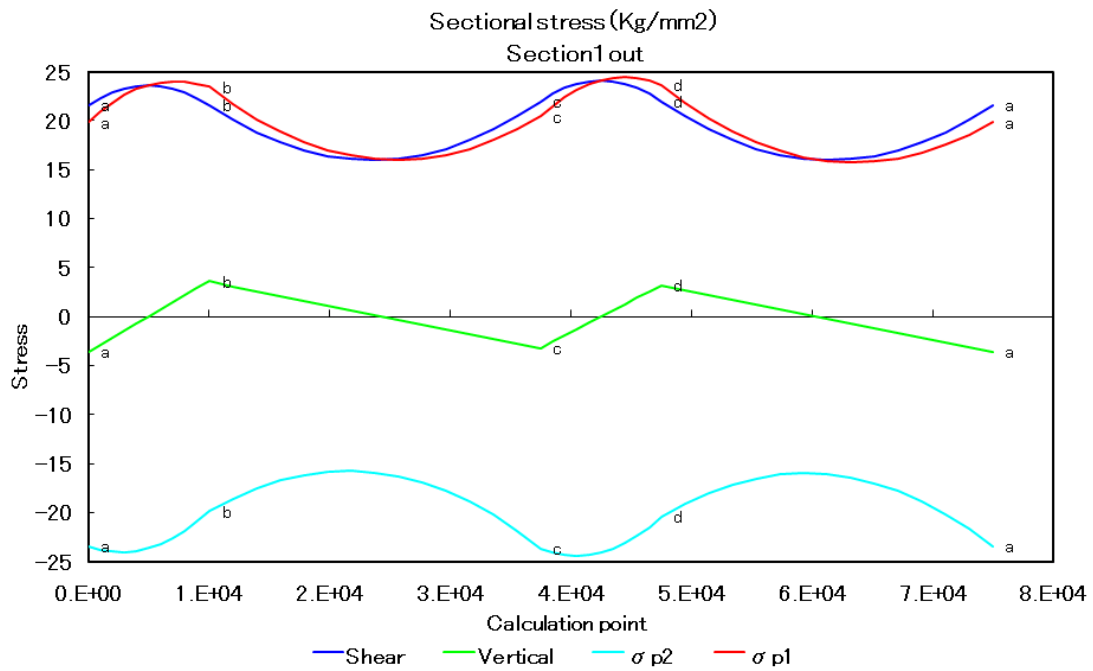


Fig.-3-11 Sectional stress of the bending-torsion theory (Rectangular section)

( 2 ) Section 2in

Fig.-3-12 shows sectional stress of the simple torsion theory and Fig.-3-13 shows that of the bending-torsion theory. Lateral axis on the graphs is girth length along the section shown on Fig.-3-1 and the sign a, b, c and d in the graph correspond to pink colored points on the figure.

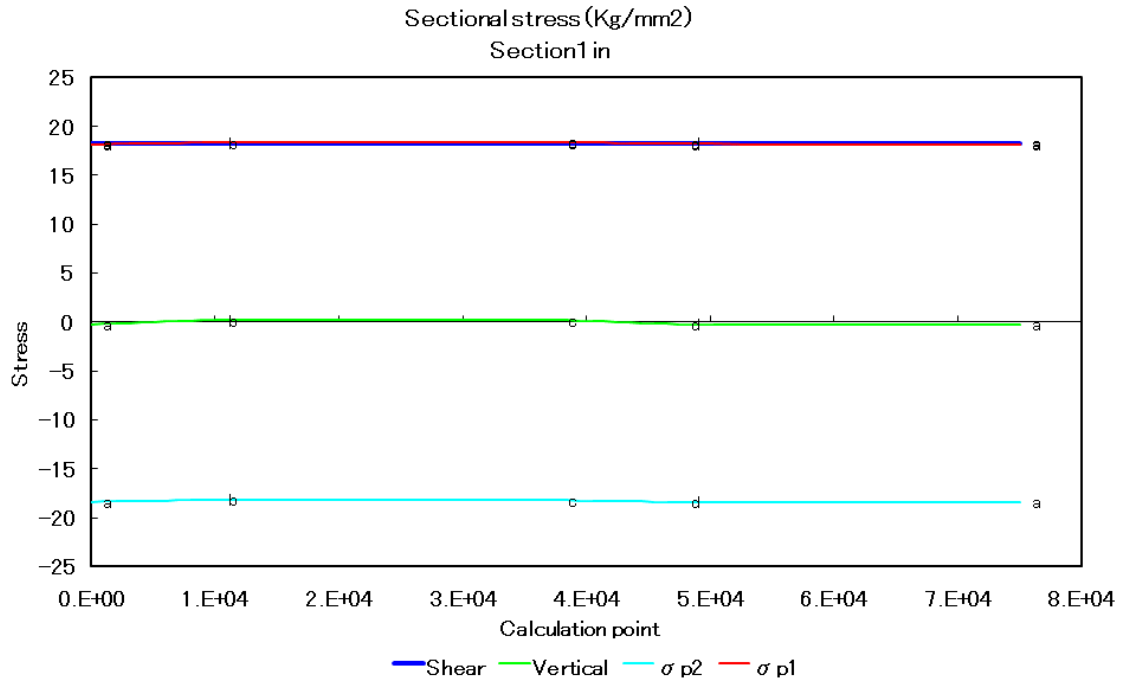


Fig.-3-12 Sectional stress of the simple torsion theory (Rectangular section)

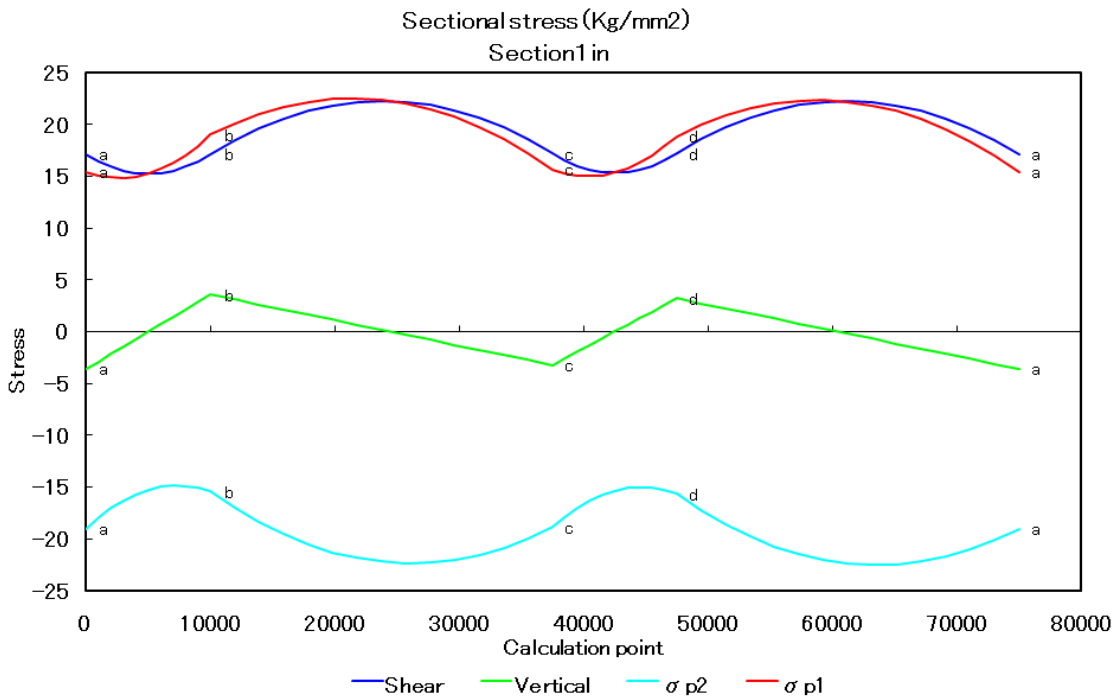


Fig.-3-13 Sectional stress of the bending-torsion theory (Rectangular section)

**Explanation of Fig.-3-12 (simple torsion):** Shear is abbreviation of shearing stress which corresponds to the sum of simple torsion stress and shearing stress relating to bending. Vertical is abbreviation of vertical stress which corresponds the sum of stresses by bending around X and Y axes.  $p_1$  and  $p_2$  are principal stresses calculated from the shearing stress and the vertical stress. Stress status of the section is almost in pure shearing because absolute value of  $p_1$  and  $p_2$  is almost equal to the shearing stress.

**Explanation of Fig.-3-13 (bending-torsion):** Shear is abbreviation of shearing stress which corresponds to the sum of simple torsion stress, bending-torsion stress and shearing stress relating to bending. Vertical is abbreviation of vertical stress which corresponds the sum of vertical stress by sectional warping and stresses by bending around X and Y axes.  $p_1$  and  $p_2$  are principal stresses calculated from the shearing stress and the vertical stress. Stress status of the section is almost in pure shearing because absolute value of  $p_1$  and  $p_2$  is almost equal to the shearing stress. The warping stress which corresponds to majority of the vertical stress may affect sectional stress to certain extent but is not dominant. The maximum shearing stress increase rate of bending-torsion versus simple torsion is about 22 %. Increase of stress occurs between b and c and d and a (side plates), and decrease occurs between a and b (bottom plate) and c and d (top ).

3 . 2 . 2 . 2 Stress in the vicinity of a free end

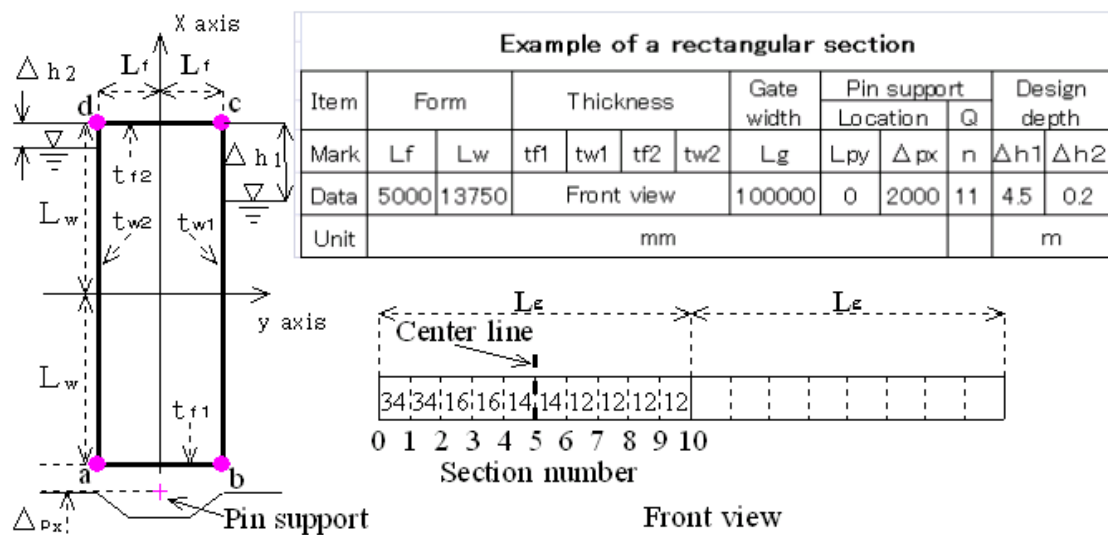


Fig.-3-14 Example of a rectangular section

Sectional stress in the vicinity of a free end of the case shown on Fig.-3-14 is presented. Applied twisting moment at a free end is comparatively small. The section number 0 in the figure is a support end and section number 10 is a free end.

( 1 ) Section 8out

Fig.-3-15 shows sectional stress of the simple torsion theory and Fig.-3-16 shows that of the bending-torsion theory. Lateral axis on the graphs is girth length along the section shown on Fig.-3-14 and the sign a, b, c and d in the graph correspond to pink colored points on the figure.

**Explanation of Fig.-3-15 (simple torsion):** Shear is abbreviation of shearing stress which corresponds to the sum of simple torsion stress and shearing stress relating to bending. Vertical is abbreviation of vertical stress which corresponds the sum of stresses by to bending around X and Y axes. Resultant is abbreviation of resultant stress which is calculated according to the shearing strain energy theory (Mises-Hencky-Huber-Theory).  $p_1$  and  $p_2$  are principal stresses calculated from the shearing stress and the vertical stress. Stress status of the section is almost in pure shearing because absolute value of  $p_1$  and  $p_2$  is almost equal to the shearing stress.

**Explanation of Fig.-3-16 (bending-torsion):** Shear is abbreviation of shearing stress which corresponds to the sum of simple torsion stress, bending-torsion stress and shearing stress relating to bending. Vertical is abbreviation of vertical stress which corresponds the sum of vertical stress by sectional warping and stresses by to bending around X and Y axes. Resultant is abbreviation of resultant stress which is calculated according to the shearing strain energy theory (Mises-Hencky-Huber-Theory).  $p_1$  and  $p_2$  are principal stresses calculated from the shearing stress and the vertical stress. Stress status of the section is almost in pure shearing because absolute value of  $p_1$  and  $p_2$  is almost equal to the shearing stress. The warping stress which corresponds to majority of the vertical stress may affect sectional stress to certain extent but is not dominant. The maximum shearing stress increase rate of bending-torsion versus simple torsion is about 114 %. Increase of stress occurs between b and c and d and a (side plates), and decrease occurs between a and b (bottom plate) and c and d (top plate).

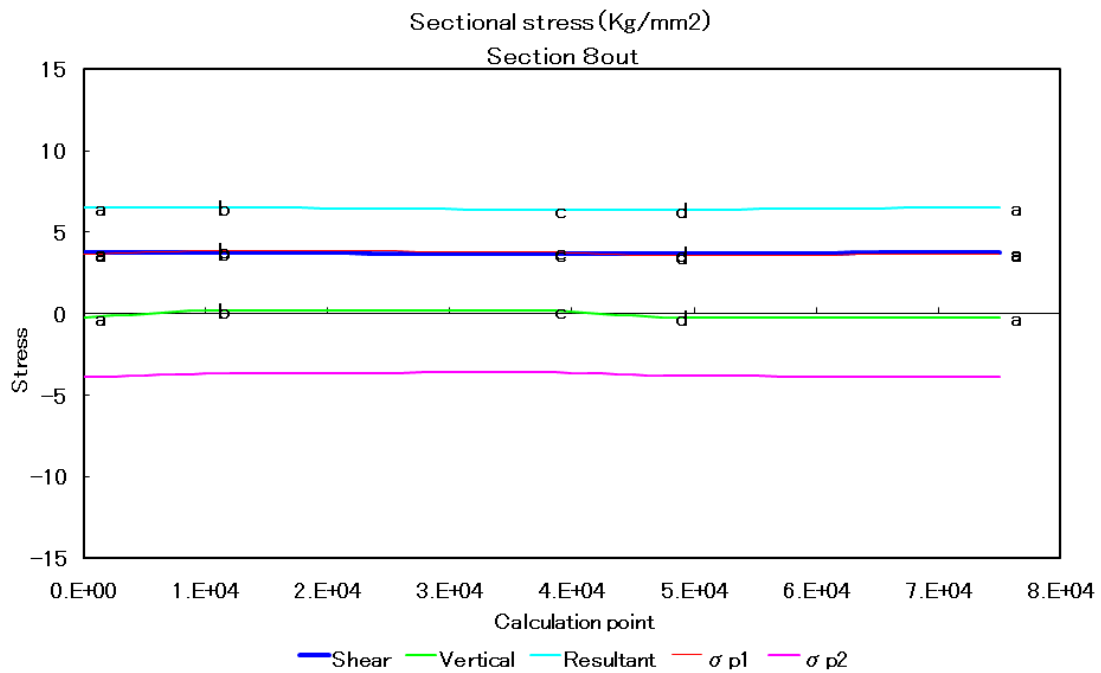


Fig.-3-15 Sectional stress of the simple torsion theory (Rectangular section)

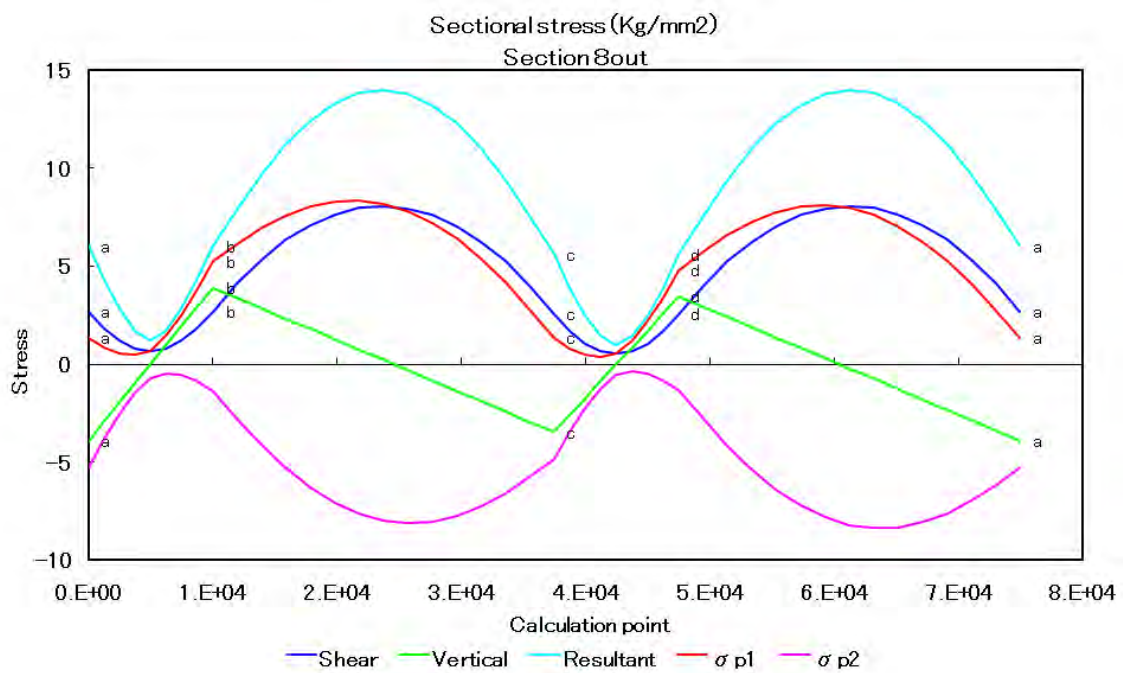


Fig.-3-16 sectional stress of the bending-torsion theory (Rectangular section)

( 2 ) Section 9in

Fig.-3-17 shows sectional stress of the simple torsion theory and Fig.-3-18 shows that of the bending-torsion theory. Lateral axis on the graphs is girth length along the section shown on Fig.-3-14 and the sign a, b, c and d in the graph correspond to pink colored points on the figure.

**Explanation of Fig.-3-17 (simple torsion):** Shear is abbreviation of shearing stress which corresponds to the sum of simple torsion stress and shearing stress relating to bending. Vertical is abbreviation of vertical stress which corresponds the sum of stresses by bending around X and Y axes. Resultant is abbreviation of resultant stress which is calculated according to the shearing strain energy theory (Mises-Hencky-Huber-Theory).  $p_1$  and  $p_2$  are principal stresses calculated from the shearing stress and the vertical stress. Stress status of the section is almost in pure shearing because absolute value of  $p_1$  and  $p_2$  is almost equal to the shearing stress.

**Explanation of Fig.-3-18 (bending-torsion):** Shear is abbreviation of shearing stress which corresponds to the sum of simple torsion stress, bending-torsion stress and shearing stress relating to bending. Vertical is abbreviation of vertical stress which corresponds the sum of vertical stress by sectional warping and stresses by bending around X and Y axes. Resultant is abbreviation of resultant stress which is calculated according to the shearing strain energy theory (Mises-Hencky-Huber-Theory).  $p_1$  and  $p_2$  are principal stresses calculated from the shearing stress and the vertical stress. Stress status of the section is almost in pure shearing because absolute value of  $p_1$  and  $p_2$  is almost equal to the shearing stress. The warping stress which corresponds to majority of the vertical stress may affect sectional stress to certain extent but is not dominant. The maximum shearing stress increase rate of bending-torsion versus simple torsion is about 86 %. Increase of stress occurs between a and b (bottom plate) and c and d (top plate), and decrease occurs between b and c and d and a (side plates).

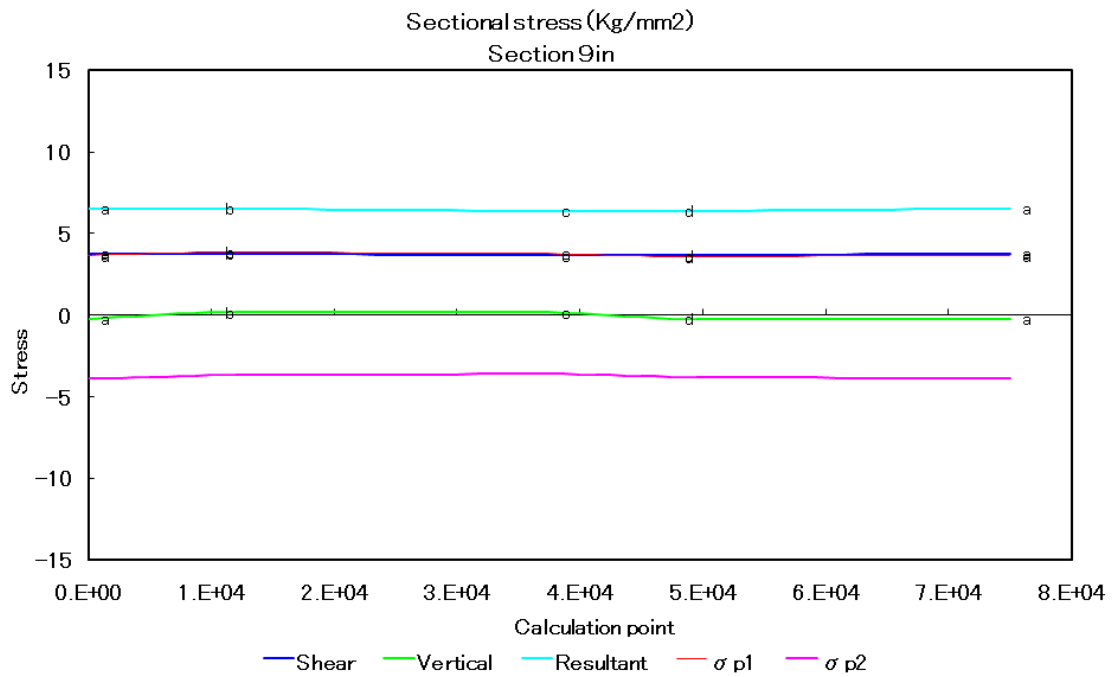


Fig.-3-17 Sectional stress of the simple torsion theory (Rectangular section)

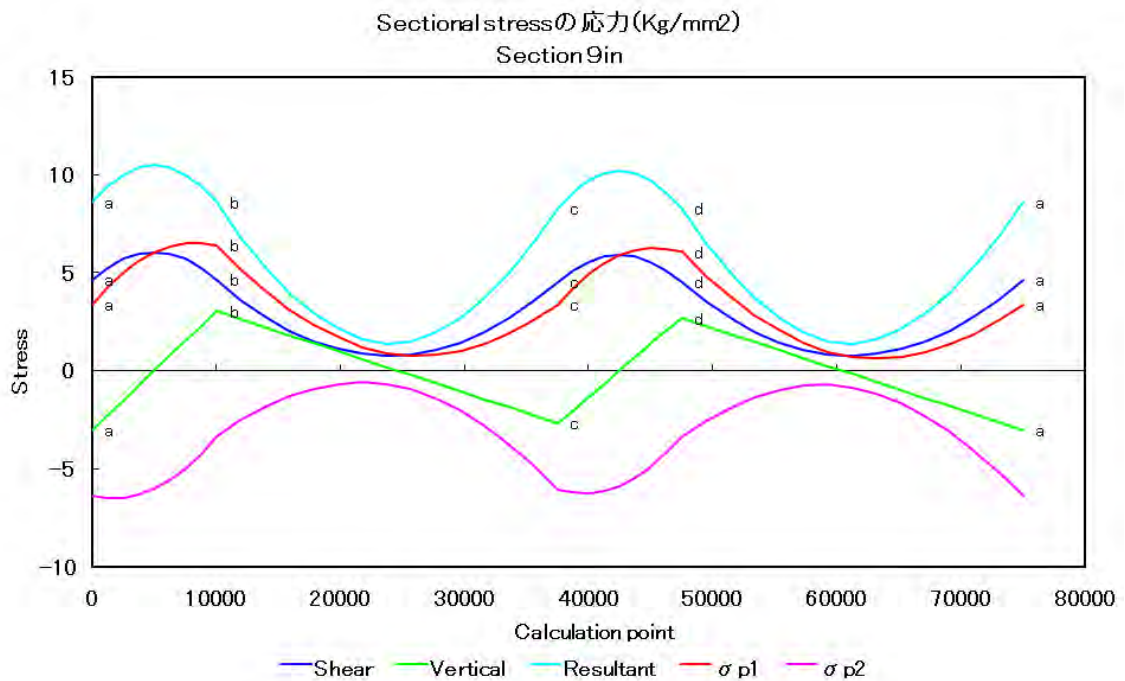


Fig.-3-18 sectional stress of the bending-torsion theory (Rectangular section)

#### 4 . Mitigation of warping

It is shown in the previous article that shearing stress on a section meanders with the existence of bending-torsion and its peak value remarkably increases sometimes. A main cause of the meander is the shearing stress which constitutes bending-torsion moment of the section. Since this stress is in equilibrium with vertical stress of sectional warping, it is estimated that the vertical stress as well as the stress meandering will vanish if there is no warping.

Stress is in proportion to the product of a form coefficient, deformation and a spring constant\*<sup>1</sup>. If a form coefficient reduces by sectional warping reduction, both shearing and vertical stress of bending-torsion may decrease or vanish (patent pending). On the other hand, the form coefficient reduction will result in section modulus decrease. Because intensity of deformation is equal to internal force divided by the product of a section modulus and a spring constant\*<sup>2</sup>, the section modulus decrease will bring out deformation increase which may cancel the effect of the form coefficient reduction.

The most important section modulus controlling deformation of a torsion type structure is  $J \tau$ \*<sup>3</sup>. The product of  $J \tau$  and a spring constant is a sectional rigidity which resists simple torsion moment. Although  $J \tau$  decreased according to the form coefficient reduction, a gate weight cut is possible by making up the lost  $J \tau$  with help of cross section form change.

The object function of optimum design is construction cost including fabrication. Although the purpose of the form coefficient operation was a gate weight cut, an optimum design of torsion type structure including this operation is possible.

##### 4 . 1 Condition of zero warping

Fig.-4-1 shows s axis which is set along a center line of thin shell making up a closed section shown in the figure where  $d_s$  is a small quantity at arbitrary point on s, t is

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\*1Contents of each item are explained at (1) of Appendix 1 Technical key factors.

\*2Contents of each item are explained at (1) of Appendix 1 Technical key factors.

\*3Relation between  $J \tau$  and  $J \tau$ , and warping, and vertical stress and  $J \tau$  and shearing stress are explained at (1) and (3) of Appendix 1 Technical key factors.

thickness of shell at the point,  $S$  is a shear center of the closed section and  $r_s$  is distance between the shear center of closed section and a tangent at the arbitrary point. And the area surrounded by  $s$  axis is denoted by  $A_s$ .

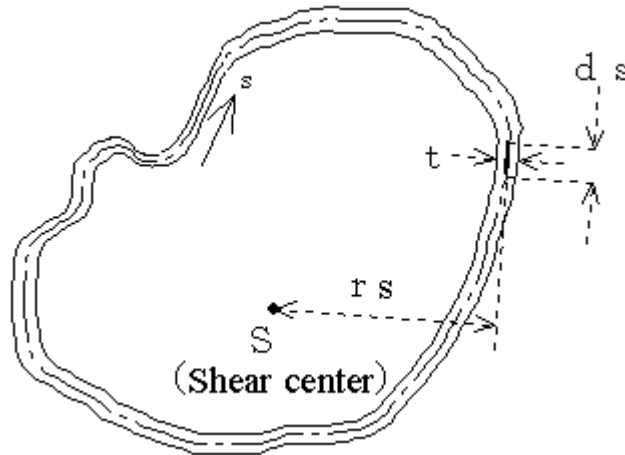


Fig.-4-1 The  $s$  axis on a closed section of thin shell

Warping function of a closed section of thin shell in Fig.-4-1 is given by below formula.

$$\Psi = \Psi_0 - \int_0^s r_s ds + 2A_s \int_0^s \frac{1}{t} ds + \int_0^s \frac{ds}{t}$$

$\Psi_0$  of above formula is a warping constant which is equal to  $\Psi$  at integration start point and is given by below formula.

$$\Psi_0 = \left( \int_0^s t \int_0^s r_s ds ds - 2A_s \int_0^s \frac{ds}{t} + \int_0^s t \int_0^s \frac{1}{t} ds ds \right) \div \int_0^s t ds$$

Warping function and a warping constant  $\Psi_0$  are going to be zero, when formula (1) below is satisfied.

$$t \times r_s = \text{constant on each section} = C \quad \dots \dots (1)$$

Sectional warping disappears when  $\Psi_0$  is zero, and vertical stress which is in proportion to  $\Psi$  and bending-torsion shearing stress which is in equilibrium with the vertical stress are also zero. In short, **formula (1) is a condition of zero warping.**

#### 4 . 2 Examples of zero warping

Two examples, a rectangular section and a lens section which is a kind of fish belly section, are presented. Formula (1) is applied to achieve a zero warping status.

##### 4 . 2 . 1 A lens section

Fig.-4-2 shows scantlings of a closed lens section of thin shell. As a shearing center coincides with a section center, formula (1) is transformed into formula (2).  $\beta_0$  is an angle made by  $r_0$  and line  $o_i$  and its range is  $0$  to  $\pi$ .  $\beta_0$  is equal to  $t_i / t_0$  where  $t_i$  is a shell thickness which is a function of  $\beta_0$  and satisfies the zero warping condition.  $s_i$  is a line  $s_i$ .

$$\beta_0 = (r_0 - s_i) \div [r_0 - s_i \times \cos(\beta_0)] \dots\dots(2)$$

A warping function and a shear flow of bending-torsion for the case of Fig.-4-2 are shown first then a sectional form which satisfies the zero warping condition is shown.

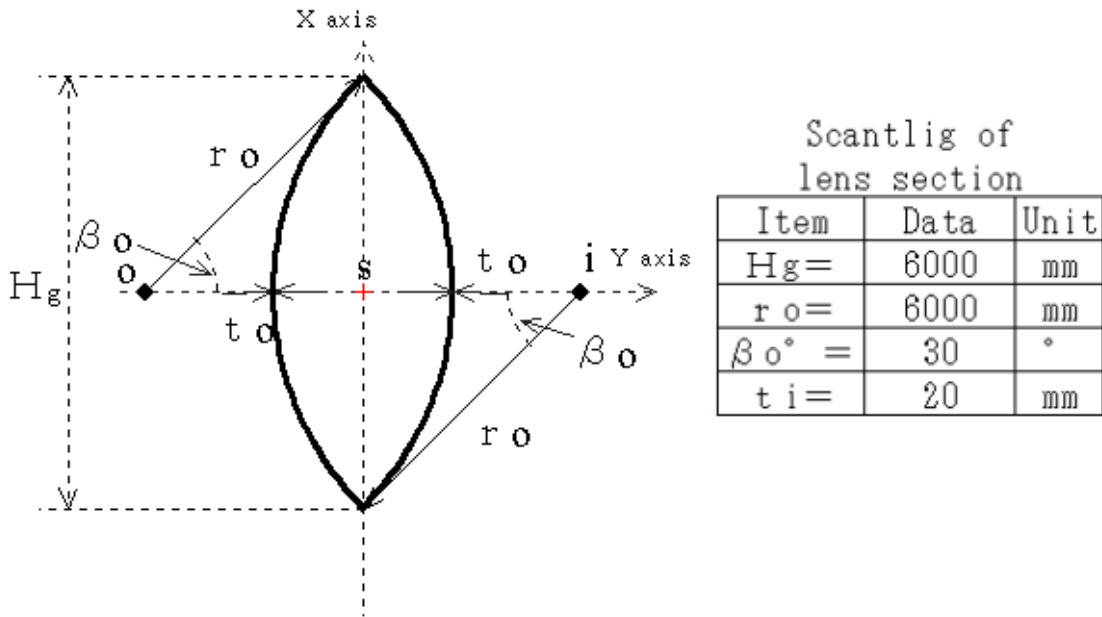


Fig.-4-2 Lens section

Fig.-4-3 is a warping function  $\beta_0$  and a shear flow of bending-torsion for a lens section shown on Fig.-4-2.  $\beta_0$  is a form coefficient of warping and vertical stress and the shear flow divided by  $t_0$  is a form coefficient for shearing stress of bending-torsion. Distributions of warping and stresses are in proportion to these graphs.

Shear flow of bending-torsion and warping function  $\Psi$  (out side of section+)  
 $H=6\text{ m}$ ,  $r_o=6.0\text{ m}$ ,  $r_i=6.0\text{ m}$ ,  $t_o=20\text{ mm}$ ,  $t_i=20\text{ mm}$

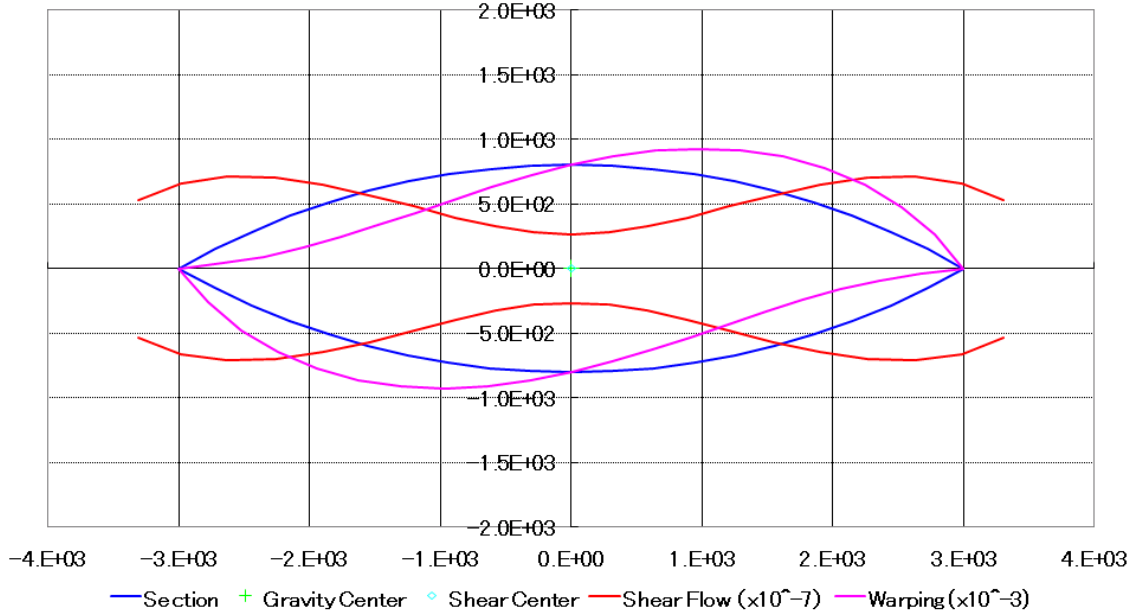
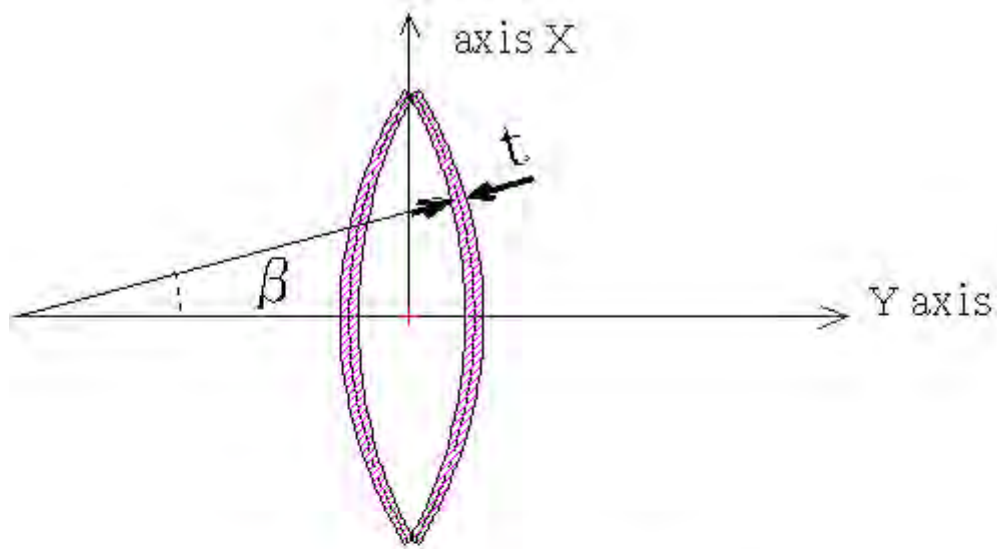


Fig.-4-3 Warping function and shear flow of bending-torsion (lens section)

Fig.-4-4 shows a form of lens section which satisfies a zero warping condition. Calculated points of formula (2) are joined together by curve lines and the thickness is shown in 10 times scale. The calculation was made on eleven points which distribute uniformly over a quarter of the section. The shell thickness on the Y axis is  $t_o$  which is original thickness of Fig.-4-2. The bending-torsion was completely excluded and shear flow and warping shown on Fig.-4-3 vanished.



Zero warping plate thickness  
(Lens section)

NO	$\beta^\circ$	$\beta$ rad.	$\eta$	t (mm)
0	0	0.000	100.0%	20.0
1	3	0.052	99.1%	19.8
2	6	0.105	96.6%	19.3
3	9	0.157	92.6%	18.5
4	12	0.209	87.6%	17.5
5	15	0.262	81.9%	16.4
6	18	0.314	76.0%	15.2
7	21	0.367	70.0%	14.0
8	24	0.419	64.1%	12.8
9	27	0.471	58.7%	11.7
10	30	0.524	53.6%	10.7

Fig.-4.4 Lens section which satisfies a zero warping condition

#### 4 . 2 . 2 A rectangular section

A section of Fundamental case shown on Fig.-3-1 is modified step by step until warping on the section disappears completely. A state of decrease in warping and torsion stress on the section is presented in below two ways.

- 1) Decrease in warping function and shear flow on a section
- 2) Change in factors of warping and results of structural analysis

Step by step method of Fundamental section approaching a zero warping condition is explained specifically, before the state of decrease in warping is presented. Zero warping is realized by plate thickness selection as was seen at a lens section. Fundamental section is shown on Fig.-4-5 after setting  $t_{f1} = t_{f2} = t_f$  and  $t_{w1} = t_{w2} = t_w$  on Fig.-3-1. As a shearing center coincides with an axial center, zero warping condition formula (1) is transformed into formula (3).

$$t_f \times L_w = t_w \times L_f \quad \cdot \cdot \cdot \cdot (3)$$

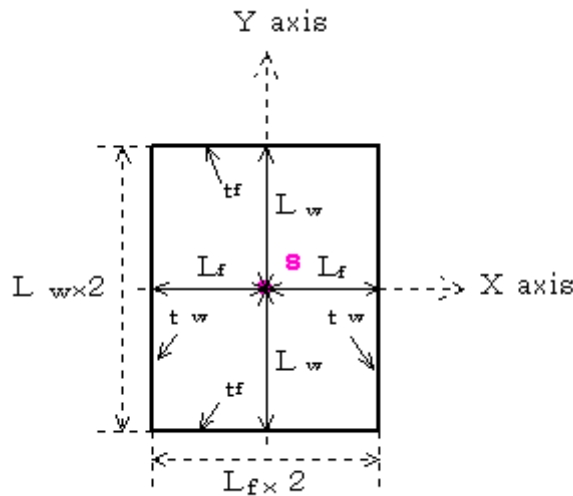


Fig.-4-5 Fundamental section (Axial center = shear center)

There are two choices to realize a zero warping condition, one is to increase  $t_w$ , another is to decrease  $t_f$ . Suppose purpose of a zero warping condition is gate weight saving, step by step approaching method is to decrease  $t_f$  step by step until  $t_f$  arrives at  $t_w \times L_f \div L_w$  12.4 mm (a zero warping condition).

( 1 ) Warping function and shear flow

Fig.-4-6 thru Fig.-4-9 show a state of decrease in warping function and shear flow. Fig.-4-6 is Fundamental section ( $t_f=34$  mm). Fig.-4-7 is a section of  $t_f=16$  mm, and Fig.-4-8 is  $t_f=14$ mm. Fig.-4-9 is a zero warping section ( $t_f=12.4$  mm).

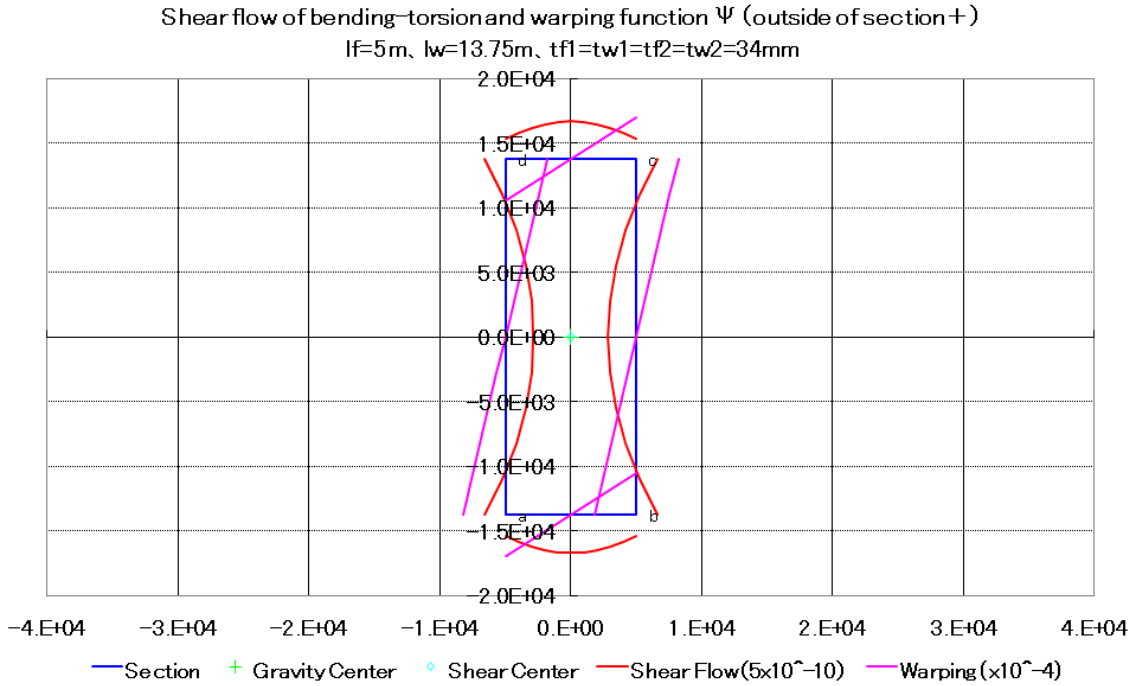


Fig.-4-6 Warping function and shear flow of bending-torsion  
(Fundamental section:  $t_f=34$  mm)

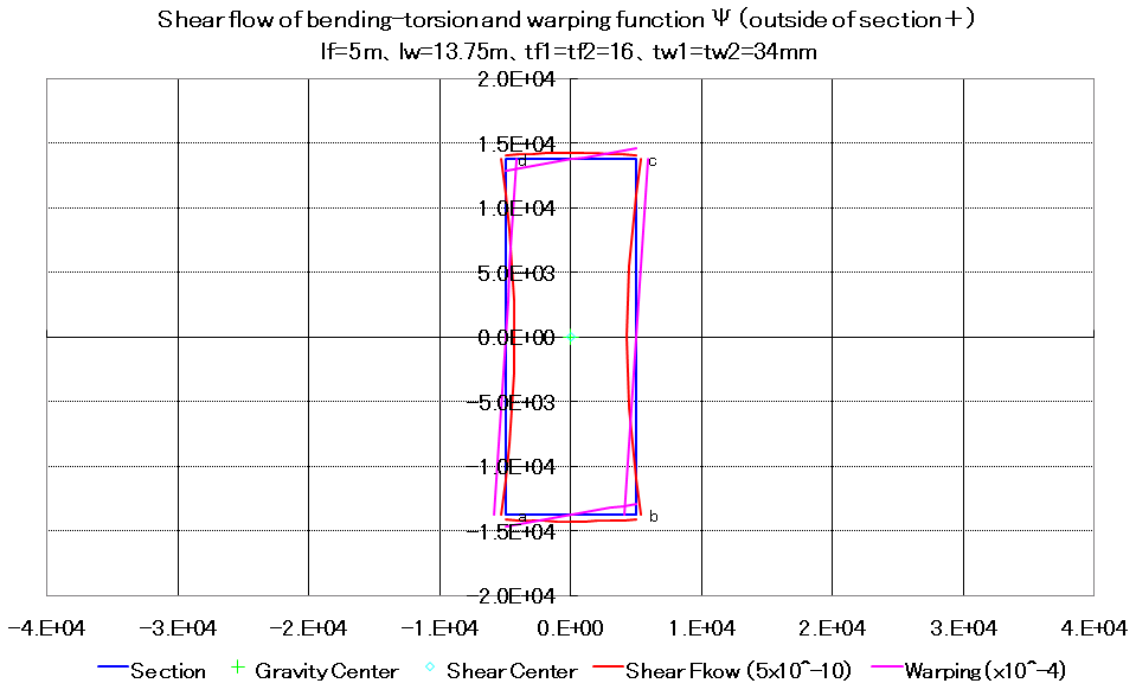


Fig.-4-7 Warping function and shear flow of bending-torsion ( $t_f=16$  mm)

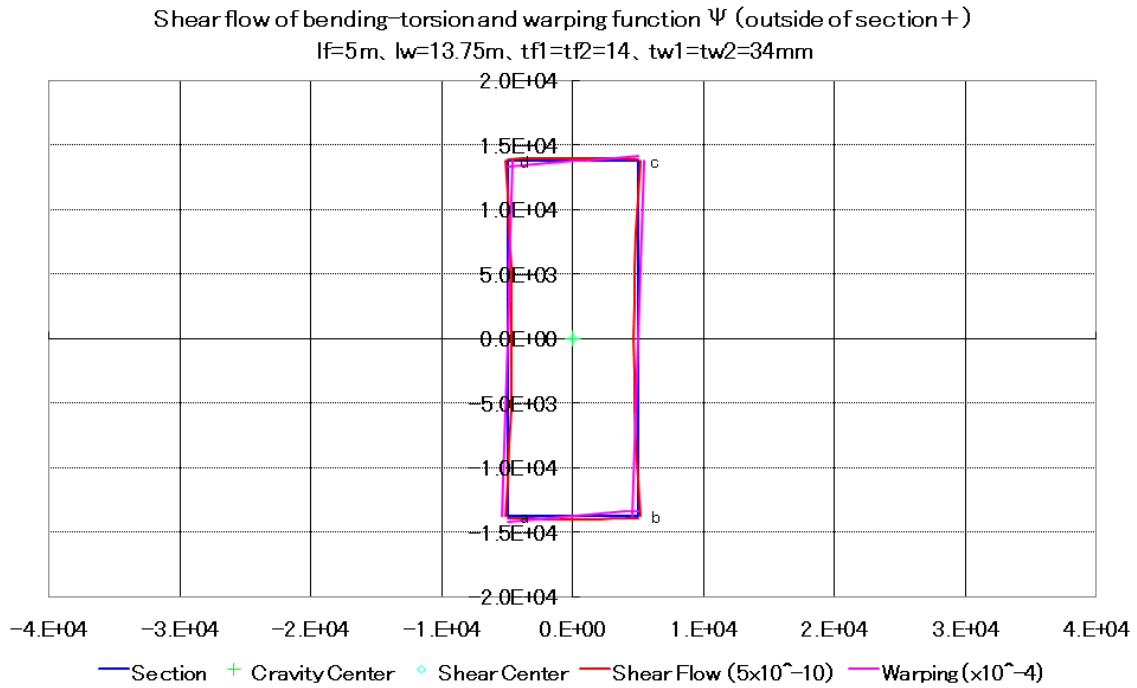


Fig.-4-8 Warping function and shear flow of bending-torsion ( $t_f=14$  mm)

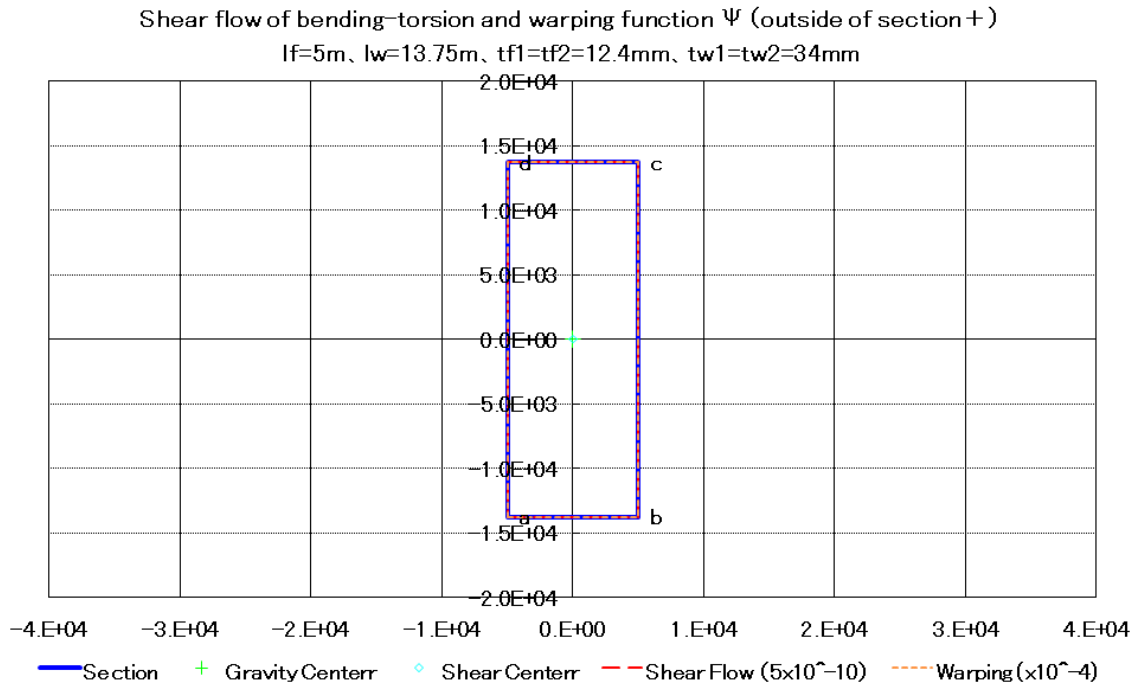


Fig.-4-9 Warping function and shear flow of bending-torsion  
 (Zero warping section:  $t_f=12.4$  mm)

( 2 ) Factors of warping and structural analyses

Purpose of this item is as follows.

- 1) To show quantitatively a status of decrease in sectional warping.
- 2) To show influence on the whole structure of the zero warping operation.

For a reference, the structural analysis was carried out according to the elastic equation method of bibliography (6).

Fig.-4-10 shows change in factors of warping and influence on structural analysis. The lateral axis is plate thickness  $t_f$ , and the vertical axis is factors of warping and results of structural analysis whose indication is in % of Fundamental case.

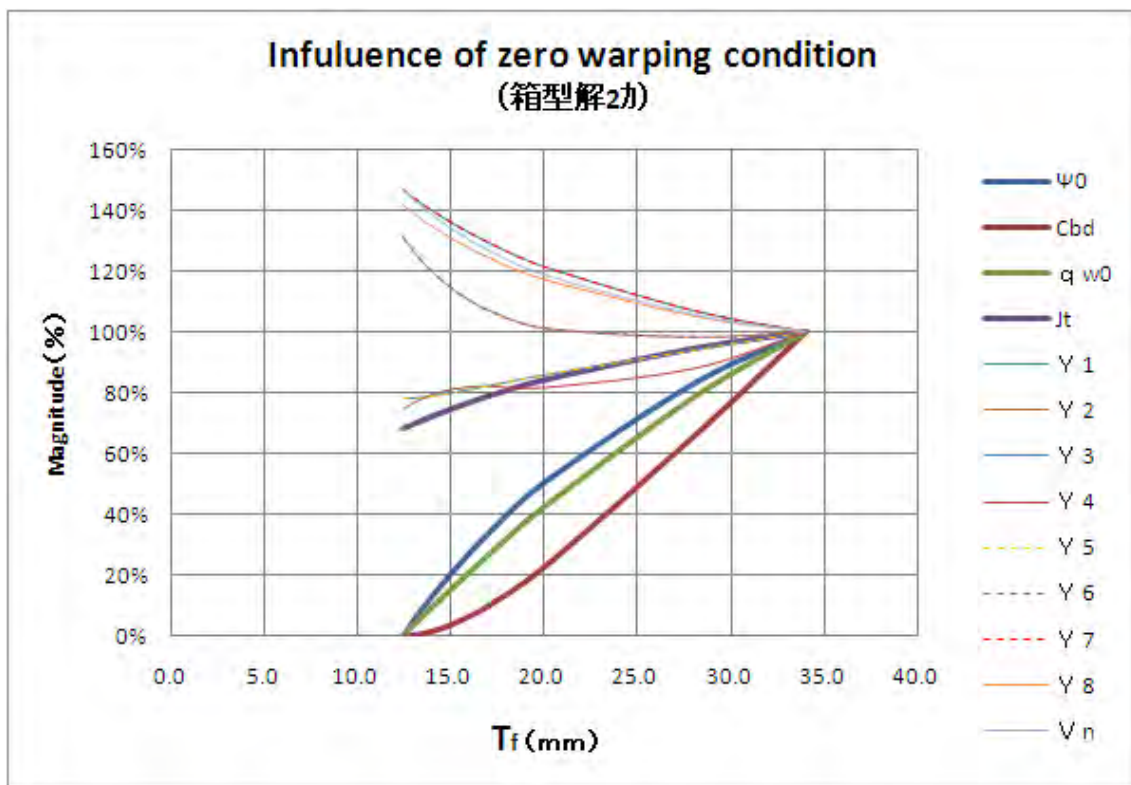


Fig.-4-10 Change in factors of warping and influence on structural analysis

**Explanation of factors of warping:** A warping constant  $\psi_0$  which relates to warping magnitude and a shear flow constant  $q_{w0}$  (value at integration starting point of shear flow  $q_w$ ) which relates to shearing stress of bending-torsion are rapidly approaching 0 at  $t_f=12.4$ mm. Two section moduli,  $C_{bd}$  (bending-torsion) and  $J_t$  (simple torsion), are also decreasing. Decrease of  $J_t$  is very important because  $J_t$  is a leading factor in deformation control of a torsion type structure. Although  $J_t$  decrease due to zero warping operation is not unreasonable, there is a possibility of

cancelling the form coefficient reduction effort because of an increase in twisting angle of the structure.

**Explanation of structural analysis:** Results of structural analysis shown on the figure are  $V_n$  (displacement in Y direction of a shear center at a free end), and  $Y_1$  thru  $Y_8$  (statically indeterminate reaction forces in Y direction of pin supports). As a result of the zero warping operation, following are observed.

- 1) A remarkable increase of  $V_n$  (due to a big increase of  $\theta$ )
- 2)  $Y_3$  thru  $Y_5$  of span center portion increased and others of span end portions decreased.
- 3) Above statement suggests an increase of bending deformation, which is still secondary matter.

#### 4 . 3 Optimum design

It is shown that the form coefficient reduction will result in section modulus decrease which will bring out stress increase due to increase of twisting angle  $\theta$ . As the stress increase can be controlled by making up the lost  $J_t$  with help of a cross sectional form change, an optimum design including the zero warping condition is possible.

##### 4 . 3 . 1 Method of a sectional form change

Although there are two choices,  $L_f$  and  $t_w$ , as a measure of the form change to increase  $J_t$ , the latter does not meet the purpose which is a gate weight cut. Rate of  $J_t$  increasing due to  $L_f$  increase is much more than rate of the form coefficients ( $Q_{w0}$  and  $Q_{w0}$ ) increasing and rate of 2nd and third derivation of  $J_t$  decreasing due to  $J_t$  increase is much more than decreasing rate of first derivation of  $J_t$ . Accordingly, it is concluded that the  $L_f$  increase is an appropriate measure of reducing bending-torsion stress.

A step by step method of  $t_f$  approaching the zero warping condition was introduced at 4 . 2 . 2. Let  $t_f$  at each step be denoted by  $t_{f \text{ step}}$ . A step by step method is also introduced for the a cross sectional form change and, in order to maintain a distance from a zero warping condition,  $t_f$  at each step is rounded out in mm scale after it is calculated by formula (4).

$$t_f = t_{f \text{ step}} \times L_{f \text{ increased}} \div 5000 \text{ (mm)} \quad \cdot \cdot \cdot \cdot (4)$$

where,  $L_{f \text{ increased}}$ : Increased  $L_f$

#### 4 . 3 . 2 Method of optimum design

Although purpose of a zero warping condition is gate weight reduction, that of optimum design is cost cut.

As cost composing factors include material, fabrication, transportation, site construction, maintenance, operation etc., minimum gate weight can not necessarily be minimum cost. For instance, there is an option that a high tensile steel plate of specially ordered thickness is fit in the shell of stress increased zone to keep minimum gate weight. But it may be better idea in cost mind to increase gate weight in order to use ordinary strength steel plate of commercial thickness since cost of material and fabrication in the option is more expensive.

So far, stress which yields corresponding to whole structural deformation due to simple torsion, bending-torsion, warping, bending etc. is considered in the study, a torsion type structure which is planned according to a zero warping condition may not be a minimum weight after stability of structural parts or stress which yields corresponding to partial deformation such as bending due to applied hydraulic pressure on gate plates and their stiffeners, bending due to reaction forces of pin supports and support ends etc. are considered.

Suppose an actual conventional measure to find out an optimum design in cost is to select best one among multiple plans, an optimum design should be selected from not only a zero warping point but also a surface made by the two lines one of which is an approaching line to the zero point and other is a cross sectional form change line to make up the lost  $J \tau$ .

Fig.-4-11 and Fig.-4-12 show image of the optimum design procedure. Fig.-4-11 is a design flow chart and Fig.-4-12 is a selection table of the zero warping condition. A purpose of the selection table is to impose the zero warping condition on the design flow and the table is expected to be a guide of decreased shell thickness setting at  $S_2$ . This table set is supposed to be prepared for each design condition and has been corresponding to the example of a rectangular section shown on Fig.-3-14.

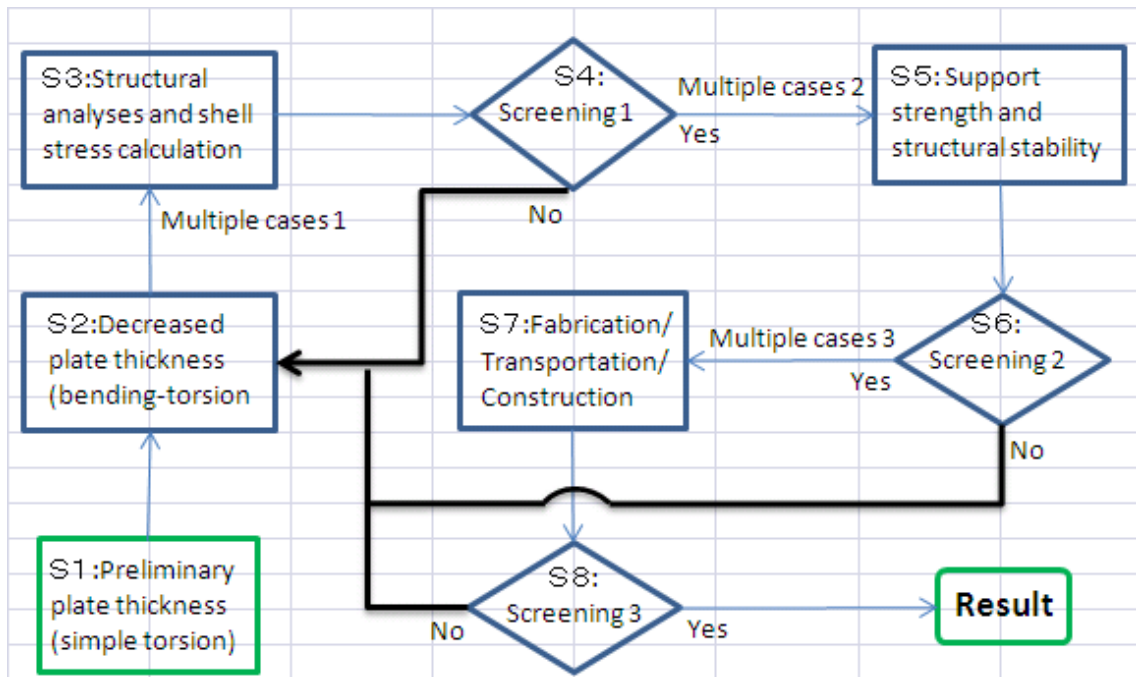


Fig.-4-11 Image of a design flow chart

Preliminary thickness: 34 mm( $t_f$ ) (Red: zero warping)														Preliminary thickness: 34mm(Weight %)(Red: zero warping)																
$L_f$ (mm)	$t_f$ step (mm)													$L_f$ (mm)	$t_f$ step (mm)															
	12.4	13	14	15	16	17	18	20	22	24	26	28	30		34	12.4	13	14	15	16	17	18	20	22	24	26	28	30	34	
5000	12.4	13	14	15	16	17	18	20	22	24	26	28	30	34	83	84	84	85	86	87	87	89	91	92	94	95	97	100		
5100	12.6	14	15	16	17	18	19	21	23	25	27	29	31	35	83	85	85	86	87	88	89	90	92	93	95	97	98	101		
5200	12.9	14	15	16	17	18	19	21	23	25	28	30	32	36	84	85	86	86	87	88	89	90	92	94	96	98	99	103		
5300	13.1	14	15	16	17	19	20	22	24	26	28	30	32	37	84	85	86	87	87	89	90	92	93	95	97	98	100	104		
5400	13.4	15	16	17	18	19	20	22	24	26	29	31	33	37	85	86	87	88	89	89	90	92	94	95	98	100	101	105		
5500	13.6	15	16	17	18	19	20	22	25	27	29	31	33	38	85	86	87	88	89	90	91	92	95	97	98	100	102	106		
5600	13.8	15	16	17	18	20	21	23	25	27	30	32	34	39	85	87	87	88	89	91	92	94	95	97	100	101	103	108		
5900	14.6	16	17	18	19	21	22	24	26	29	31	34	36	41	87	88	89	90	91	93	94	96	97	100	102	105	107	111		
6700	16.6	18	19	21	22	23	25	27	30	33	35	38	41	46	91	92	93	95	96	98	100	102	105	108	110	113	116	122		
7500	18.5	20	21	23	24	26	27	30	33	36	39	42	45	51	95	97	98	100	102	104	105	109	112	116	119	123	126	133		
Preliminary thickness: 16 mm( $t_f$ ) (Red: zero warping)														Preliminary thickness: 16mm(Weight %)(Red: zero warping)																
$L_f$ (mm)	$t_f$ step (mm)													$L_f$ (mm)	$t_f$ step (mm)															
	5.8			6	7	8	9	10	11	12	13	14	15		16	5.8			6	7	8	9	10	11	12	13	14	15	16	
5000				6	7	8	9	10	11	12	13	14	15	16						87	88	90	92	93	95	97	98	100		
5100						9	10	11	12	13	14	15	16	17						89	90	92	94	95	97	99	101	102		
5200						9	10	11	12	13	14	15	16	17						89	91	92	94	96	98	99	101	103		
5300					8	9	10	11	12	13	14	15	16	17					87	89	91	93	95	96	98	100	102	103		
5400					8	9	10	11	12	13	15	16	17	18					88	90	91	93	95	97	100	102	104	106		
5500					8	9	10	11	13	14	15	16	17	18					88	90	92	94	97	99	101	103	105	106		
5600					8	9	11	12	13	14	15	16	17	18					88	90	94	96	98	99	101	103	105	107		
5900					8	9	10	11	12	13	15	16	17	18	19				89	91	93	95	97	99	103	105	107	109	111	
6700	7.8			9	10	11	13	14	15	17	18	19	21	22					93	96	98	102	105	107	111	114	116	120	122	
7500	8.7			9	11	12	14	15	17	18	20	21	23	24					96	101	103	108	111	116	118	123	126	131	133	
Preliminary thickness: 14 mm( $t_f$ ) (Red: zero warping)														Preliminary thickness: 14mm(Weight %)(Red: zero warping)																
$L_f$ (mm)	$t_f$ step (mm)													$L_f$ (mm)	$t_f$ step (mm)															
	5.1				5	6	7	8	9	10	11	12	13		14	5.1				5	6	7	8	9	10	11	12	13	14	
5000								8	9	10	11	12	13	14							89	90	92	94	96	98	98	100		
5100								8	9	10	11	12	13	14	15						89	91	93	95	97	99	101	102		
5200								8	9	10	11	12	13	14	15						89	91	93	95	97	99	101	103		
5300								8	9	10	11	12	13	14	15						89	92	94	96	98	100	102	104		
5400								8	9	10	11	12	13	15	16						90	92	94	96	98	100	104	106		
5500								8	9	10	11	13	14	15	16						90	92	94	96	101	103	105	107		
5600								8	9	11	12	13	14	15	16						90	93	97	99	101	103	105	107		
5900						8	9	10	11	12	13	15	16	17						91	94	96	98	100	103	107	109	112		
6700						9	10	11	13	14	15	17	18	19						96	99	101	107	109	112	117	119	122		
7500	7.6				8	9	11	12	14	15	17	18	20	21						96	99	105	108	113	116	122	125	130	133	
Preliminary thickness: 12 mm( $t_f$ ) (Red: zero warping)														Preliminary thickness: 12mm(Weight %)(Red: zero warping)																
$L_f$ (mm)	$t_f$ step (mm)													$L_f$ (mm)	$t_f$ step (mm)															
	4.4					5	6	7	8	9	10	11	12		4.4					5	6	7	8	9	10	11	12			
5000										8	9	10	11	12									91	93	96	98	100			
5100										8	9	10	11	12	13								91	94	96	98	101	103		
5200										8	9	10	11	12	13								92	94	96	99	101	103		
5300										8	9	10	11	12	13								92	95	97	99	102	104		
5400										8	9	10	11	12	13								93	95	97	100	102	105		
5500										8	9	10	11	13	14								93	95	98	100	105	108		
5600										8	9	11	12	13	14								93	96	101	103	106	108		
5900								8	9	10	11	12	13	15									94	97	100	102	105	107	113	
6700								9	10	11	13	14	15	17									100	103	106	112	115	118	124	
7500	6.5					8	9	11	12	14	15	17	18										100	103	110	113	120	123	130	133

Fig.-4-12 Image of a selection table of the zero warping conditions

## 5 . Conclusion

- (1) A torsion type closed thin shell section is overwhelmingly superior to a bending type structure.
- (2) There are two types of structural torsion, the simple torsion and the bending-torsion.
- (3) Sectional shearing stress distribution of simple torsion moment is almost uniform.
- (4) The sectional shearing stress distribution meanders by additional bending-torsion moment.
- (5) In some case, peak stress of additional bending-torsion is much more than 200% of simple torsion alone.
- (6) Stress is in proportion to a form coefficient  $\times$  deformation  $\times$  a spring constant.
- (7) Form coefficient can be diminished by eliminating sectional warping.
- (8)  $J t$  will reduce due to warping reduction operation, and stress and deformation increase accordingly.
- (9) Control of stress increase is possible by making up the lost  $J t$  with help of a cross sectional form change.
- (10) An optimum design including the zero warping condition is possible.

## Appendix 1 . Technical key factors of a torsion type structure

( 1 ) Factors of stress and deformation (their contents and relations)

- 1) Stress = Form coefficient  $\times$  Deformation  $\times$  Spring constant
- 2) Deformation = Internal force  $\div$  Sectional rigidity
- 3) Sectional rigidity = Section modulus  $\times$  Spring constant

Table - 1 Factors of stress (contents)

Stress	Form coefficient	Deformation	Spring constant
$\sigma_b$ (bending)	$x, y$	$\ddot{x}, \ddot{y}$	E
$\tau_b$ (bending)	$\frac{Q_{yz}}{t}, \frac{Q_{xy}}{t}$	$\dddot{x}, \dddot{y}$	E
$\tau_s$ (simple torsion)	$\frac{Q_z}{t}$	$\dot{\theta}$	G
$\sigma_z$ (bending-torsion)	$\psi$	$\ddot{\theta}$	E
$\tau_w$ (bending-torsion)	$\frac{Q_w}{t}$	$\ddot{\theta}$	E

Table - 2 Factors of deformation (contents)

Deformation	Internal force	Sectional rigidity	
		Section modulus	Spring constant
$\ddot{x}, \ddot{y}$	$mby, mbx$	$I_y, I_x$	E
$\dddot{x}, \dddot{y}$	$Q_x, Q_y$	$I_y, I_x$	E
$\dot{\theta}$	$T_s$	$Jt$	G
$\ddot{\theta}$	(?)	$(Jt \times C_{bd})^{0.5}$	$(G \times E)^{0.5}$
$\ddot{\theta}$	$T_w$	$C_{bd}$	E

( 2 ) Relations of stress to  $t$  and  $S$  (degree of effect).

Red colored factors in Table - 1 and Table - 2 are affected from  $t$  and  $S$ . Table - 3 shows degree of  $t$  and  $S$  effect on section moduli. Table - 4 shows degree of  $t$  and  $S$  effect on stress. Effect degree of deformation is invert of section modulus and effect degree of stress is a product of form coefficient and deformation. Effect degree analysis is based upon geometric similarity of sectional form and of  $t$  distribution.

Table - 3 Degree of  $t$  and  $S$  effect on section modulus

Section modulus	Effect degree of $t$ and $S$
$I_y, I_x$	$t^1 \cdot s^3$
$J_t$	$t^1 \cdot s^3$
$(J_t \times C_{bd})^{0.5}$	$t^1 \cdot s^4$
$C_{bd}$	$t^1 \cdot s^5$

Table - 4 Degree of  $t$  and  $S$  effect on stress

Stress	Effect degree of $t$ and $S$		
	Form coefficient	Deformation	Stress
$\sigma_b$ (bending)	$S$	$t^{-1} \cdot s^{-3}$	$t^{-1} \cdot s^{-2}$
$\tau_b$ (bending)	$S^2$	$t^{-1} \cdot s^{-3}$	$t^{-1} \cdot s^{-1}$
$\tau_s$ (simple torsion)	$S$	$t^{-1} \cdot s^{-3}$	$t^{-1} \cdot s^{-2}$
$\sigma_z$ (bending-torsion)	$S^2$	$t^{-1} \cdot s^{-4}$	$t^{-1} \cdot s^{-2}$
$\tau_w$ (bending-torsion)	$S^2$	$t^{-1} \cdot s^{-5}$	$t^{-1} \cdot s^{-3}$

( 3 ) Factors of deformation (their contents and relations)

1) Displacement = Deformation  $\times$  Form coefficient

Table - 5 Factors of displacement (contents)

Displacement		Deformation	Form coefficient
Direction	Mark		
X direction	x	x	1
Y direction	y	y	1
X direction	u	$\theta$	$-(y-L_p y)$
Y direction	v	$\theta$	$x-L_p x$
Warping (Z)	w	$\dot{\theta}$	$\Psi$

## Appendix 2 . Bibliography

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